

1 Energy and exergy analysis of microchannel central solar receivers for pressurised fluids

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3 D. D'Souza^{a,b}, M.J. Montes^{a,b}, M. Romero^a and J. González-Aguilar^{a,&}

4
5 ^a High Temperature Processes Unit, IMDEA Energy, Avda. Ramón de la Sagra 3,
6 28935, Móstoles, Spain

7 ^b E.T.S. Ingenieros Industriales - UNED, C/Juan del Rosal 12, 28040 Madrid, Spain

8
9 [&]Corresponding author: J. Gonzalez-Aguilar, jose.gonzalez@imdea.org

10 11 **Abstract**

12 Within the new generation of advanced central solar receivers, microchannel
13 pressurised gas receivers are emerging as reliable and efficient alternatives to operate at high
14 temperatures and pressures. This paper presents an optimisation and comparative analysis of
15 different compact plate-fin type structures, constituting the receiver's absorber panels, classified
16 according to the type of fin arrangement inside: plain rectangular, plain triangular, wavy, offset
17 strip, perforated, and louvred fin. A versatile thermo-fluid receiver model is implemented,
18 allowing simple variation of characteristic geometric parameters of each structure. Exergy
19 efficiency is chosen as the optimisation function, as it considers both heat and pressure losses.

20 The framework of the analysis is set by the receiver's boundary conditions, operating at
21 the design point conditions of a solar thermal power plant. For each compact structure, the
22 optimal configuration is determined, providing interesting findings that have not been reported
23 in the state-of-the-art to date. Although all geometries show good thermal performance, the
24 perforated and plain rectangular configurations demonstrate the best exergy efficiencies of
25 59.21% and 58.80%, respectively, favouring taller and narrower channels. This analysis
26 methodology could be seamlessly extrapolated to other gases and working conditions, owing to
27 the thermo-fluid model's versatility, to reveal the optimal configuration for each case.

28 29 **Keywords**

30 Microchannel; pressurised fluids; solar receiver; solar thermal power; energy efficiency;
31 exergy efficiency.

33 **Acronyms**

34	CHE	Compact Heat Exchanger
35	CST	Concentrated Solar Thermal
36	CSP	Concentrated Solar Power
37	CFD	Computational Fluid Dynamics
38	EA	Electrically Assisted
39	EDM	Electrical Discharge Machining
40	HCE	Heat Collector Element
41	HTF	Heat Transfer Fluid
42	HX	Heat Exchanger
43	IRENA	International Renewable Energy Agency
44	LCOE	Levelized Cost of Electricity
45	LF	Louvred Fin
46	LMTD	Log Mean Temperature Difference
47	NIST	National Institute of Standards and Technology
48	NREL	National Renewable Energy Laboratory
49	OSF	Offset Strip Fin
50	PCHE	Printed Circuit Heat Exchanger
51	PF	Perforated Fin
52	PGR	Pressurised Gas Receiver
53	PFHE	Plate Fin Heat Exchanger
54	PHE	Plate Heat Exchanger
55	PRF	Plain Rectangular Fin
56	PTF	Plain Triangular Fin
57	RNM	Resistance Network Model
58	RPC	Reticulated Porous Ceramic
59	sCO ₂	Supercritical Carbon Dioxide
60	SCR	Solar Central Receiver
61	SHE	Spiral Heat Exchanger
62	SLM	Selective Laser Melting
63	STPP	Solar Thermal Power Plant
64	TRM	Thermal Resistance Model
65	WF	Wavy Fin
66		
67		

68 **NOTATION**

69 Latin letters

70	A	Area (m^2)
71	B	Breadth (m)
72	c_p	Specific heat at constant pressure ($\text{J kg}^{-1} \text{K}^{-1}$)
73	D	Diameter (m)
74	F	View factor
75	f_D	Darcy pressure friction loss factor
76	f_F	Fanning pressure friction loss factor
77	h	Specific enthalpy (J kg^{-1})
78	H	Height (m)
79	h_{conv}	Convection heat transfer coefficient ($\text{W m}^{-2} \text{K}^{-1}$)
80	j	Colburn factor
81	k	Thermal conductivity ($\text{W m}^{-1} \text{K}^{-1}$)
82	L	Length (m)
83	M	Mass (kg)
84	\dot{m}	Mass flow rate (kg s^{-1})
85	N	Number of channels/elements
86	Nu	Nusselt number
87	ΔP	Pressure Drop (Pa)
88	p	Pitch (m)
89	P	Pressure (Pa)
90	Pr	Prandtl number
91	\dot{Q}	Thermal power (W)
92	r	Radius (m)
93	R	Thermal Resistance (K W^{-1}), Ideal gas constant ($\text{J kg}^{-1} \text{K}^{-1}$)
94	Re	Reynolds number
95	t	Channel wall thickness (m)
96	T	Temperature (K)
97	U	Overall heat transfer coefficient ($\text{W m}^{-2} \text{K}^{-1}$)
98	v	Velocity (m s^{-1})
99	V	Volume (m^3)
100		
101		Greek Letters
102	η	Efficiency
103	ρ	Density (kg m^{-3})
104	μ	Dynamic viscosity (Pa s)

105	Δ	Differential
106	δ	half angle of the cone subtended by the sun's disc (rad)
107		
108		Subscripts
109	0	Base case
110	abs	Absorbed
111	amb	Ambient
112	ap	Aperture
113	avg	Average
114	b	Base wall
115	c	Channel
116	cond	Conduction
117	conv	Convection
118	e	Element, electrical
119	h	Hydraulic, horizontal
120	in	Inlet
121	loss	Loss
122	net	Net
123	opt	Optical
124	out	Outlet
125	p	Pressure, plate
126	rad	Radiation
127	rec	Receiver
128	ref	Reflection
129	th	Thermal
130	v	Vertical
131	w	Wall
132		

133 **1. Introduction**

134 According to IRENA, the weighted average Levelized Cost Of Electricity (LCOE) of
135 Concentrating Solar Power (CSP) has decreased between 2010 and 2020 from USD 0.34/kWh_e
136 to USD 0.108/kWh_e [1]. This LCOE reduction over the last decade has been mainly due to the
137 lowering of CSP installation costs, which in 2020 became 50% cheaper than in 2010, owing to
138 greater economies of scale. Nevertheless, if the target of USD 0.06/kWh_e is to be achieved [2],
139 it is essential to increase the global thermal performance of Solar Thermal Power Plants
140 (STPPs). To this end, the Gen3 CSP Roadmap established three development pathways for
141 central receiver technology on the basis of the Heat Transfer Fluid (HTF) employed: molten
142 salts, particles, and gas-phase fluids [3].

143 Within these three pathways, the technology most developed and commercialised is
144 based on molten salts. An overarching objective for the next generation of CSP plants, across all
145 receiver development pathways, is an increase in the receiver outlet temperature, from the
146 conventional 565 °C to 700 °C, to increase power cycle efficiency [4],[5],[6]. This, however,
147 brings significant technical challenges in molten salts, mainly related to chemical instability and
148 material corrosion [7]. Regarding particle receivers, they can stably operate at high
149 temperatures, up to 1,000 °C, and inherently permit direct storage, but their main drawback lies
150 in the downstream primary heat exchanger between the particles and the working fluid in the
151 power cycle. Particle conveyance, attrition and transport also remain a challenge for these
152 receivers [8],[9]. At last, gas-phase receivers can operate at high temperatures (>1,000 °C) in
153 general, and they are stable across a wide temperature range besides being cheaper, less
154 corrosive than commercial molten salts and non-environmentally hazardous [10]. However,
155 there are several challenges to its widespread adoption including difficulties in thermal storage,
156 higher pressure drops (leading to larger fluid circulation power demands) and poor performance
157 when a gas is used as HTF due to its unfavourable thermo-physical properties. The poor heat
158 evacuation also limits the operation temperature of solar receivers since their solid surfaces may
159 not be cooled sufficiently [11]. Within gaseous fluid receivers, Pressurised Gas Receivers
160 (PGRs) offset some of the inherent disadvantages of gas phase receivers by ensuring that there
161 is adequate mass flow in all channels and avoiding flow instabilities characteristic of volumetric
162 receivers [12],[13],[14]. Besides, if the gas is pressurised, the pressure drop is reduced for the
163 same mass flow and cross section as density is approximately proportional to pressure, thus the
164 velocity is much lower at high pressure. Additionally, HTF is not limited to air, which is the
165 case with atmospheric gas receivers and other gases with more favourable heat transfer
166 characteristics may be used [15]. Pressurised gas receivers may be further classified on the basis
167 of the gas employed (air, helium, nitrogen, etc.), the irradiation conditions (directly, indirectly
168 or hybrid) and the flow path geometry [16]. With respect to their internal geometry, tubular
169 receivers are by far the most studied and developed [17],[18],[19], though alternative concepts

170 exist such as the embedded channel receiver [20], impinging jet receiver [21],[22],[23],
171 Reticulated Porous Ceramic (RPC) lined cavity receivers [24],[25],[26] and microchannel
172 receiver [27],[28].

173 Microchannel pressurised gas receivers, which are the focus of this study, seek to
174 achieve the objective of improving the heat transfer to the gaseous fluid by increasing the heat
175 transfer area for the same receiver volume, and improving the convection heat transfer
176 coefficient due to decreased diffusion length compared to conventional channels or tubes [29].
177 Within the microchannel receiver concept, the use of a pressurised gas is advisable to reduce the
178 pressure drop and flow instabilities, as explained previously. Microchannel receivers, as the
179 name suggests, consist of miniature channels. The inspiration for such receivers comes on the
180 back of growth in Compact Heat Exchanger (CHE) technologies (cf. Annex A.1) and their
181 increased real-world application [30],[31]. The reduction in channel size, i.e. compactness of
182 CHEs, results in a volume reduction for the same effective heat transfer area; higher heat
183 transfer coefficients; and higher pressure drop, although this effect could be offset by using
184 pressurised fluids.

185 A review into the suitability of different CHE geometries for solar receivers led to the
186 development of a microchannel receiver prototype [27],[28], which was divided into 12 parallel
187 channels with each channel being 1 mm wide and 3 mm high. Rectangular ribs were attached on
188 top of each channel. These ribs had the same width as the channel and a height and pitch of
189 2 mm and 1 mm, respectively. The first rib was placed just at the entrance of the air passage.
190 The receiver was manufactured of Inconel 625 using Selective Laser Melting (SLM). The outer
191 (or top) surface of 30 mm × 30 mm was the irradiated plane. Experiments were carried out
192 using pressurised air (2-6 bar) and resulted in a thermal efficiency of around 64% with a
193 pressure drop of around 750 mbar. Another interesting microchannel prototype based on
194 structures commonly used in a Plate Fin Heat Exchanger (PFHE) was developed and
195 numerically analysed in [32–34]. This 3 MW_{th} receiver, made of Inconel 625, consists of several
196 plates joined by diffusion with rectangular fins and square-shape channels. It was designed to
197 heat supercritical CO₂ (sCO₂) from 530 °C to 700 °C at 20 MPa. In the same work, a parametric
198 analysis was also carried out to study the effect of the hydraulic diameter, number of vertical
199 rows, and channel thickness on the thermal resistance and pressure drop. It was observed that
200 the thermal resistance was directly related to the hydraulic diameter and number of vertical
201 rows, whereas the pressure drop indirectly so. Increasing the channel thickness slightly reduced
202 the thermal resistance, but it leads to a more substantial increase in the pressure drop as the
203 receiver breadth was fixed. On a much smaller scale, absorber panels with a PFHE structure and
204 channel sizes in the order of hundreds of micrometres were manufactured using Electrical
205 Discharge Machining (EDM) [35],[36]. Haynes 230 was chosen as the receiver material. Lab-
206 scale absorber panels of 2 cm × 2 cm were made and proven to absorb 100 W cm⁻² of incident

207 flux using $s\text{CO}_2$ at $650\text{ }^\circ\text{C}$ and $80\text{-}200\text{ bar}$ as working fluid, reporting thermal efficiencies
208 around 90% . Finally, a recent $s\text{CO}_2$ receiver prototype using the PFHE concept has been
209 developed in the National Renewable Energy Laboratory (NREL). In this 2 MW_e design, the
210 compact structure consists of two attached plates with a wavy fin arrangement between them;
211 these plates act as the absorber surfaces of the concentrated solar radiation and they are
212 arranging forming a cavity [37]. Several studies have investigated the shape of similar cavity
213 receivers, optimising the geometries and configurations, with the aim of improving the
214 performance of the receiver by reducing the radiation and convection losses from its surface
215 [38],[39],[40],[41]. High fin density (up to 32 fins/cm) and thin walls were expected to improve
216 the heat transfer besides providing crucial mechanical support against high internal pressures
217 (25 MPa). The receiver, and its individual panels, were constructed of Inconel 625 and was
218 predicted to have an efficiency of around 90% . A 10 MW_e receiver based on this concept has
219 been designed and simulated. In this last case, the absorber plates were arranged to form an
220 external cylindrical receiver.

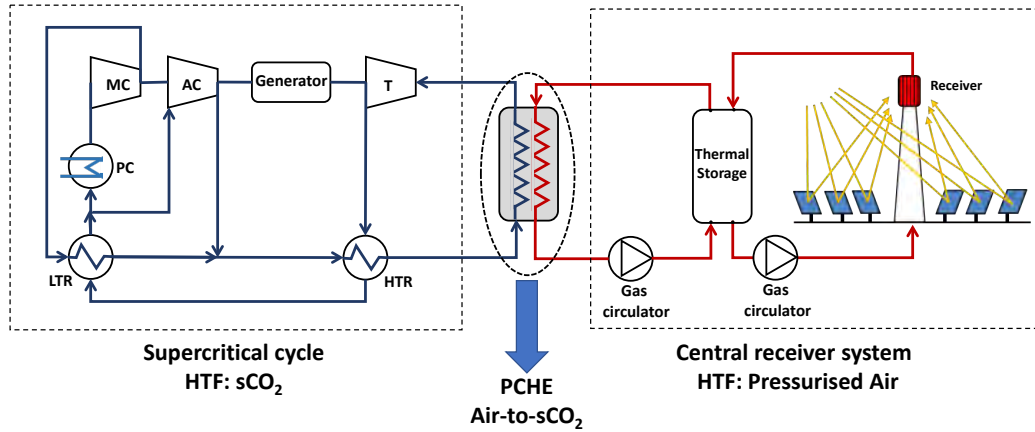
221 The state of the art shows that, to date, very few varieties of CHE sub-structures or
222 internal flow geometries have been investigated, namely the plain square/rectangular fin type
223 structure [32], the pin fin structure [35] and the wavy fin type structure [37]. However, there are
224 several other potential CHE internal flow structures that have neither been individually analysed
225 nor collectively compared in the context of their application to solar receivers. In this regard,
226 this paper aims at conducting an in-depth study of those potential geometries for microchannel
227 receivers using pressurised gas. The manuscript is organised as follows. The overall framework
228 within which the analysis is performed is laid out in section 2. For a proper comparative
229 analysis, it is imperative to set a suitable operational framework and boundary conditions for the
230 receiver. These operating conditions are detailed in section 2.1, when describing the global
231 STPP based on the microchannel pressurised gas receiver. After establishing these boundary
232 conditions, the sizing and operating conditions of the receiver subsystem are set out in section
233 2.2. The geometrical characterisation of the different compact structures, as well as the scope of
234 the parametric study, are presented in section 2.3. Having defined the global operational
235 structure, section 3 presents the thermo-fluid dynamic model developed and aspects pertaining
236 to its application in this analytical work. The numerical model used to analyse and compare the
237 various CHE geometric structures is detailed in section 3.1 followed by its validation in section
238 3.2. The selection of appropriate performance indicators is important to any comparative and
239 optimisation analysis and this is dealt with in section 3.3. Section 4 is devoted to the results
240 including an analysis of the performance and behaviour of the different receiver types in section
241 4.1, and finally a comparative analysis is elucidated in section 4.2. Exergy efficiency is
242 identified as a suitable figure of merit as it considers the exergy increase associated to the fluid
243 heat gain, and the exergy decrease caused by the pressure drop and the heat loss. The results

244 present the optimal geometric parameters for each compact structure and the comparison
245 between optima. At last, section 5 summarises the main conclusions, as well as future research
246 lines.
247

248 **2. Framework for microchannel receiver analysis**

249 *2.1. Layout and nominal conditions of solar thermal power plant*

250 The performance analysis of a solar receiver primarily requires an adequate framework,
251 which is fixed by the overall STPP performance at design point. The operating conditions of the
252 receiver are firstly imposed by the useful thermal power required by the thermal cycle.
253 Although this pressurised receiver can be coupled to several power cycles, a sCO₂ power cycle
254 has been considered following a layout similar to that depicted in Figure 1.
255



256

257 **Figure 1.** Scheme of a STPP based on a pressurised air central receiver coupled to a
258 supercritical CO₂ power cycle

259

260 As shown in Figure 1, the supercritical layout selected is the conventional
261 recompression thermodynamic cycle. The cycle power output has been set at 10 MW_e with a
262 solar multiple of 1.5. An electrical power of 10 MW_e is considered representative for an initial
263 prototype that will later be scaled to a commercial level. The isentropic efficiencies of both the
264 turbine and the compressors have been set at 92% and 88%, respectively; a dry cooling by
265 means of a pre-cooler is assumed; and the sCO₂ pressure and temperature at the turbine inlet are
266 200 bar and 688 °C, respectively. At these conditions, the nominal thermal efficiency is 49.57%,
267 thus the thermal power required in the primary heat exchanger is 20.17 MW_{th} [42]. Assuming a
268 thermal efficiency in the source heat exchanger equal to 98%, and considering the solar multiple
269 previously mentioned, the total thermal power in the central solar receiver is 30.26 MW_{th}. The
270 heat transfer fluid in the proposed receiver is pressurised air. The air temperatures at the inlet
271 and outlet of the solar receiver are also determined by the power cycle conditions. Specifically,
272 if the source heat exchanger is assumed to be balanced and the temperature difference between
273 the two fluid streams is constant and fixed at 12 °C, then the inlet and outlet air temperatures are
274 557.6 °C and 700 °C, respectively. The air pressure at the receiver inlet is taken as 25 bar. Since
275 the pressure difference between sCO₂ and pressurised air streams is high, a PCHE, which is

276 capable of operating under such conditions, is recommended for use as the HX, coupling the
 277 solar field to the power cycle with previous studies having undertaken such design studies [42].

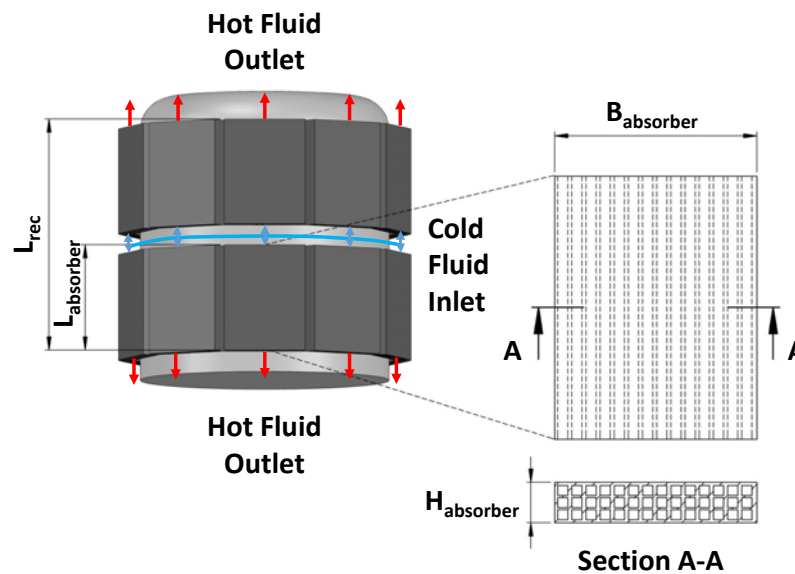
278 For the receiver simulation model, the thermodynamic properties of the pressurised air
 279 have been sourced from the NIST database [43], for temperature steps below 1.5 °C and
 280 pressure steps of roughly 1 mbar. Furthermore, assumed environmental conditions are the
 281 ambient temperature at 25 °C, the sky temperature at 15 °C and the wind speed equal to 1 m/s.

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283 *2.2. Configuration and characteristics of microchannel solar receiver*

284 Before making a performance analysis of the various compact structures, it is first
 285 necessary to define an overall receiver structure. As shown in Figure 2, an external cylindrical
 286 receiver configuration has been defined having 20 rectangular panels in a parallel configuration,
 287 uniformly irradiated, through which pressurised air flows in a single pass. These absorber panels
 288 (henceforth referred to as absorbers) are assembled so as to form 2 vertical rows of 10
 289 cylindrically arranged absorbers. CHE structures are implemented in each absorber. Cold air
 290 enters from a common inlet manifold, located between the two rows, before splitting into the
 291 individual absorbers. This configuration is similar to the one adopted for the sCO₂ receiver
 292 proposed by NREL [37].

293



294

295 **Figure 2.** Receiver configuration, i.e. external cylindrical like receiver comprised of 20 parallel
 296 rectangular absorber panels arranged cylindrically in 2 vertical rows.

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298 Inconel 617 has been selected as the receiver material because of its machinability and
 299 high temperature corrosion resistance [44],[45]. Deferring to the state of the art [46],[47], the
 300 receiver aspect ratio (receiver length to diameter ratio) is fixed at 0.7 with the maximum and

301 minimum mean incident fluxes set at 800 kW m^{-2} and 400 kW m^{-2} . These along with additional
 302 operation boundary conditions, summarised in Table 1, are used to size the absorbers and, in
 303 turn, the overall receiver in an iterative process which is detailed in Section 2.4.

304

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Table 1. Operational boundaries conditions of pressurised air receiver.

Parameter	Unit	Operating Limits	
		Minimum	Maximum
Mean incident flux	kW m^{-2}	400	800
Channel velocity	m s^{-1}	-	50
Outer surface temperature	$^{\circ}\text{C}$	-	800
Absorber temperature gradient (outer to back surfaces)	$^{\circ}\text{C}$	-	200
Reynolds number	-	10^4	-

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The temperature gradient from the absorber's external irradiated surface to the back is expected to be high given the low thermal conductivity of Inconel 617 [44] and the generally poor heat transfer characteristics of air; hence the addition of an upper limit to this temperature gradient as well as mean incident flux which are both presented in Table 1. As a consequence of air's low density, the absorber cross section area, directly related to the absorber breadth and hence receiver diameter, required for a given absorber mass flow rate and channel velocity is relatively high. Keeping this in mind, the number of absorbers in parallel (which determines the mass flow rate in each absorber) and the bounds of mean incident flux have been set to maintain a reasonable receiver aspect ratio. Given that the absorbers tend to broader and shorter dimensions, stacking the absorbers one above the other mitigates, to an extent, the low receiver aspect ratio issue by effectively doubling the receiver length.

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For the working temperatures considered in this study (above 700°C), a cavity type receiver is most recommended to reduce radiation heat losses [46]. Nevertheless, there is a recent research line that seeks to decrease the radiation losses by the reduction of the view factors, using microscopic or macroscopic geometries that would act as solar traps [8],[48]. At the microscopic scale, the external receiver proposed in this paper has adopted the configuration developed by NREL for their pressurised microchannel receiver [37], as mentioned in the state of the art review. This design employs cylindrical quartz tubes attached perpendicularly to its external surface, in such a way as to reduce the view factor and the convective losses.

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For the external receiver proposed in this work, cylindrical quartz tubes with an aspect ratio (height-to-diameter ratio) of 0.5 are considered. The view factor of this cavity is calculated using a conventional formula for parallel circular disks with centres along the same normal [49],

$$F_{rec-ap} = \frac{1}{2} \left[X - \sqrt{X^2 - 4 \left(\frac{R_2}{R_1} \right)^2} \right], \quad (1)$$

329 where $X = 1 + \frac{1+R_2^2}{R_1^2}$, $R_1 = \frac{r_1}{L}$, $R_2 = \frac{r_2}{L}$; r_1/r_2 and L are the quartz cylinder radius and length,
 330 respectively. Assuming $R_1 = R_2 = 1$, $X = 3$ and the view factor is $F_{rec-ap} = 0.382$, which is
 331 the value introduced in the program.

332 It is important to point out that this estimation of the view factor can be varied if
 333 different configurations are adopted as solar traps, the emissivity value may also be changed if
 334 different materials are considered. One of the advantages of the simulation model developed is
 335 its versatility to adapt to many designs. A summary of the proposed receiver model and its
 336 working conditions are presented in Table 2.

337

338 **Table 2.** Summary of the main thermal and geometric parameters of the pressurised air receiver.

Parameter	Unit	Value
Material	-	Inconel 617
Inlet temperature	°C	557.6
Outlet temperature	°C	700
Inlet pressure	bar	25
Mass flow rate	kg s ⁻¹	191.49
Receiver area	m ²	63.44
Receiver length	m	3.79
Receiver diameter	m	5.41

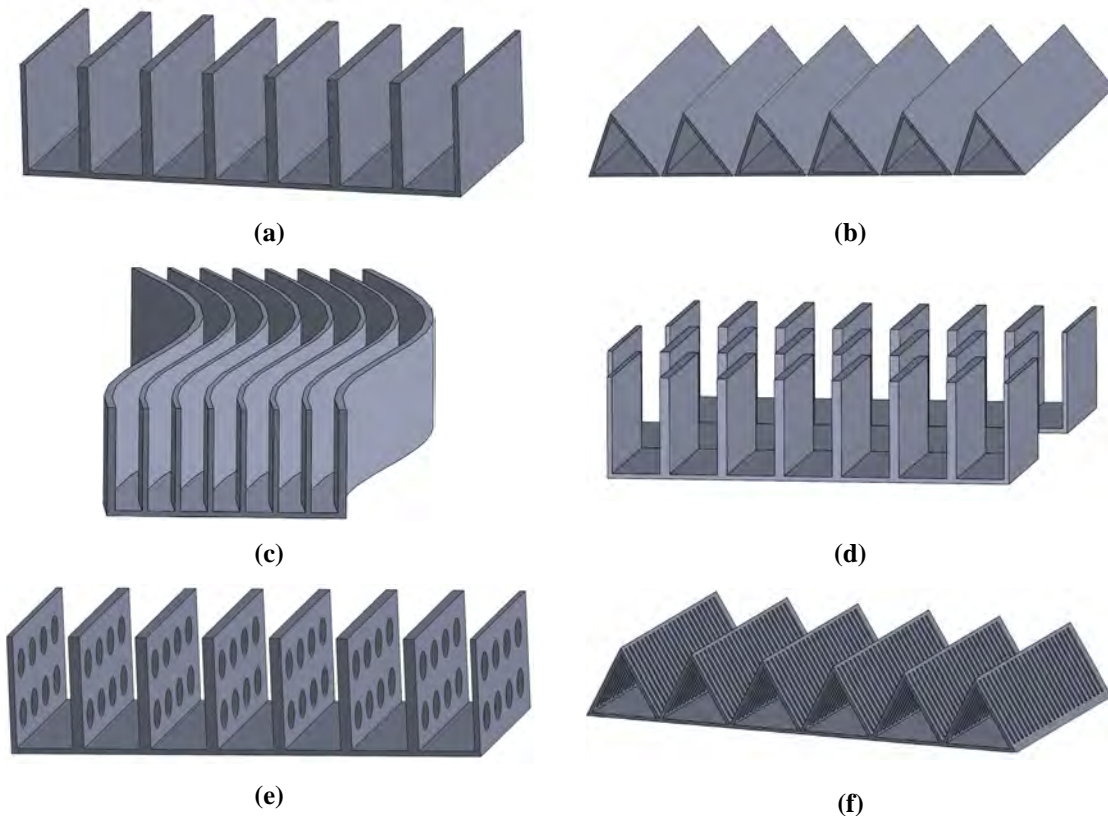
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341 2.3. Characterisation of compact structures forming flow channels of absorber panels

342 The selected CHE channel geometries, for the internal flow paths of each absorber,
343 analysed and compared in this work are the following [50]: Plain Rectangular Fin (PRF), Plain
344 Triangular Fin (PTF), Wavy Fin (WF), Offset Strip Fin (OSF), Perforated Fin (PF) with
345 rectangular cross-section, and triangular shaped Louvred Fin (LF). These geometries are
346 presented in Figure 3.

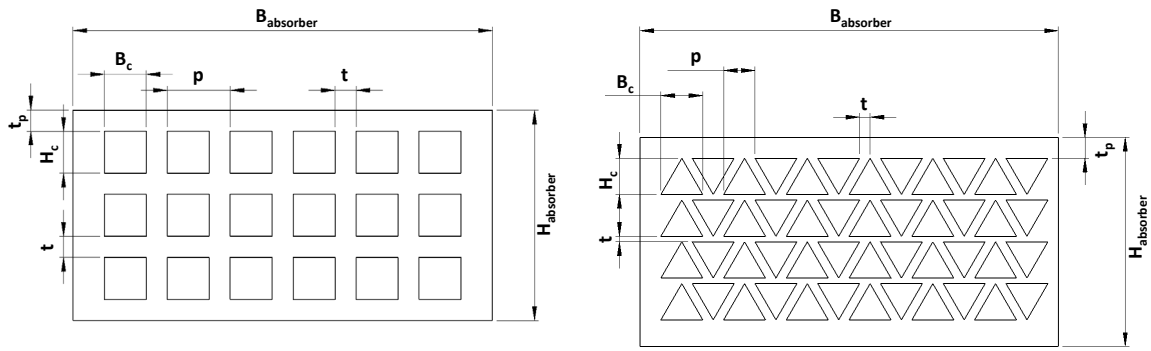
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348 **Figure 3.** CHE channel geometries analysed in the pressurised air receiver model. (a) Plain
349 Rectangular Fin (PRF); (b) Plain Triangular Fin (PTF); (c) Wavy Fin (WF); (d) Offset strip Fin
350 (OSF); (e) Perforated (Rectangular) Fin (PF); and (f) Louvred (Triangular) Fin (LF).

351

352 The channel geometries presented in Figure 3 can be characterised by common
353 parameters. Identifying these parameters and studying the effects of their variations will allow
354 for an optimisation analysis that reveals the best parameter set for each channel geometry type.
355 Besides, the analysis also facilitates a comparison among the different receivers, each with
356 different channel geometries. Four parameters, common to all channel geometries, have been
357 identified as the most crucial and these are the channel height (H_c), channel breadth (B_c),
358 channel wall thickness (t) and the number of vertical channels or rows (N_{cv}). Note that the
359 thicknesses of the walls separating horizontally and vertically adjacent channels are taken as
360 identical (t).



362 **Figure 4.** Channel geometrical parameters for rectangular fin (left) and triangular fin (right)
 363 receivers.

364

365 All the channel geometries are either rectangular or triangular shaped channels when
 366 viewed from the flow inlet/outlet. Figure 4 depicts the geometric parameters defining these
 367 structures including the previously defined constant absorber breadth ($B_{absorber}$) besides
 368 derived parameters such as the channel pitch (p) and absorber height ($H_{absorber}$), both of which
 369 are calculated after defining the four variable parameters.

370 The parametric study varying the four parameters uses the datapoints presented in
 371 Table 3 and all combinations of the variables are fed into the simulation model. It should be
 372 noted that for the LF geometry, H_c and B_c were initiated at 7 mm as excessive pressure drops
 373 were observed at lower values.

374

375 **Table 3.** Parametric study performed for optimisation analysis.

Parameter	Unit	Studied Parameter Values				
H_c	mm	6.00 ^{&}	8.25	10.50	12.75	15.00
N_{cv}	-	3	5	7	8	10
B_c	mm	6.00 ^{&}	8.25	10.50	12.75	15.00
t	mm	1.00	1.50	2.00	2.50	3.00

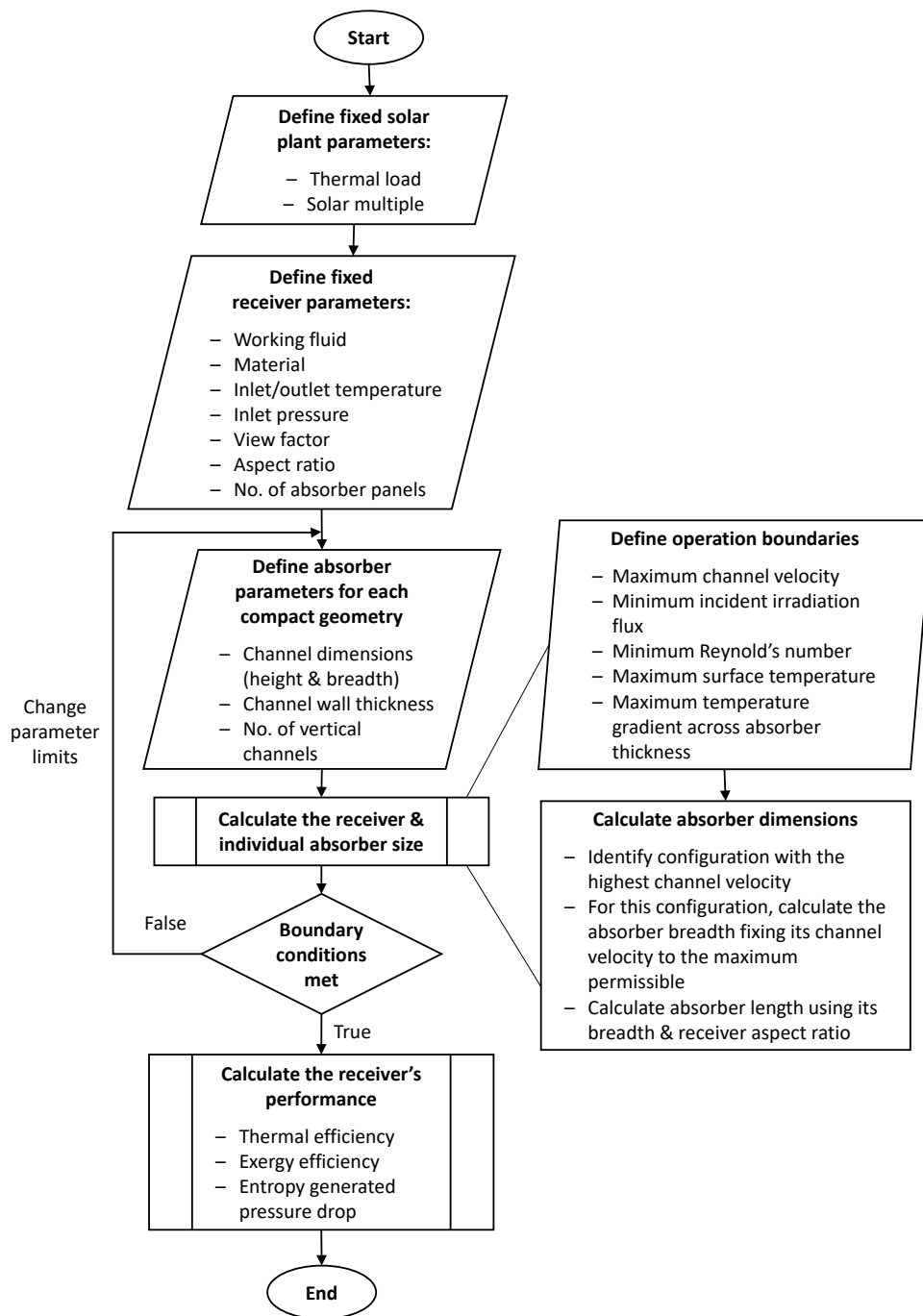
376 [&]For the LF geometry, H_c and B_c are initiated at 7 mm to avoid excessive pressure drop.

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378 **3. Numerical model: overview and method of application**

379 *3.1. Thermo-fluid dynamic model of microchannel solar receiver*

380 In order to analyse the various proposed compact structures for the pressurised receiver,
 381 a bi-dimensional model has been developed, exploiting its implicit simplicity and versatility,
 382 permitting different geometries to be easily incorporated and evaluated. The overall structure of
 383 the receiver model is portrayed in Figure 5.
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 386
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Figure 5. Overall process flow of the pressurised air receiver model.

388 After determining the global operating conditions of the receiver (Table 2), the analysis
 389 of the individual absorber is performed. It is important to note that all absorbers of the receiver
 390 are considered to operate under the same conditions and hence only one absorber need be
 391 analysed to comprehend the performance of the overall receiver.

392 The model works by dividing the absorber into multiple elements along its length, each
 393 referred to as a Heat Collector Element (HCE). For each HCE, the radial heat flux is initially
 394 assumed to be uniform and normal to every surface, and it is evaluated at the average
 395 temperature between the inlet and outlet of the HCE. These are conventional assumptions for bi-
 396 dimensional solar receiver models [51],[52].

397 The incident concentrated solar radiation (\dot{Q}_{solar}), affected by the solar trap structure,
 398 impinges and is absorbed by the absorber panel (\dot{Q}_{abs}). Most of this absorbed radiation is
 399 transmitted by conduction through the panel walls and the compact structure ($\dot{Q}_{cond,wall}$) to be
 400 finally transmitted by convection to the working fluid ($\dot{Q}_{conv,HTF}$). Since the outer wall of the
 401 panel is usually at a higher temperature than the ambient, there is a convection and radiation
 402 heat loss ($\dot{Q}_{loss,conv}$ and $\dot{Q}_{loss,rad}$, respectively). The total heat loss also includes the
 403 contribution due to reflected radiation ($\dot{Q}_{loss,ref}$), which is not absorbed by the panel. The solar
 404 trap arrangement seeks to reduce these heat losses. These heat transfer phenomena are
 405 summarised in equations (2)-(5):

$$\dot{Q}_{solar} = \dot{Q}_{abs} + \dot{Q}_{loss,ref} , \quad (2)$$

$$\dot{Q}_{abs} = \dot{Q}_{cond,wall} + \dot{Q}_{loss,conv} + \dot{Q}_{loss,rad} , \quad (3)$$

$$\dot{Q}_{cond,wall} = \dot{Q}_{conv,HTF} , \quad (4)$$

$$\dot{Q}_{loss} = \dot{Q}_{loss,conv} + \dot{Q}_{loss,rad} + \dot{Q}_{loss,ref} . \quad (5)$$

406

407

408 This system of four equations is completed by a first law energy balance applied to the working
 409 fluid, as it passes through each HCE as expressed in equation (6);

$$\dot{Q}_{conv,HTF} = \dot{m} \left[(h_{out} - h_{in}) + \frac{1}{2} (v_{out}^2 - v_{in}^2) \right] , \quad (6)$$

410 where \dot{m} is the mass flow rate, h is the enthalpy and v is the fluid velocity. In this equation,
 411 potential energy is neglected because, although the receiver is vertical, the height of the
 412 absorber panel is short. Additionally, the top row of the absorber panels has a different potential
 413 energy change than the bottom row as in one the flow is upwards and in the other downwards.
 414 As the model is a simplified amalgamation of all the panels, it was considered unnecessary to do
 415 take the negligible, equal and opposite potential energy changes into account. The required HTF
 416 thermal properties at the HCE outlet are calculated once the HCE inlet thermal properties and

417 the boundary conditions are known. Of course, the inlet conditions of an element are simply the
418 outlet conditions of the preceding element with the exception of the very first element whose
419 inlet conditions are predefined.

420 The outlet pressure is determined by calculating the pressure drop across the element
421 and subtracting that from the inlet pressure. With these two properties known, the remaining
422 required fluid properties can be determined. This process is sequentially implemented from the
423 first HCE (at the absorber inlet) to the final HCE (at the absorber outlet). If the HTF outlet
424 temperature is within the tolerance range of the setpoint i.e. $700\text{ }^{\circ}\text{C} \pm 3\text{ }^{\circ}\text{C}$, the performance
425 indicators of the absorber (thermal and exergy efficiency, pressure drop, etc.) are evaluated
426 before proceeding to the next absorber configuration. These process steps are schematically
427 outlined in Figure 6. Determination of the heat transfer coefficient and friction factor is required
428 in each HCE and is done by implementing empirical and semi-empirical correlations. These
429 correlations are unique to each CHE geometry and are tabulated in the annexes A.2 and A.3.

430

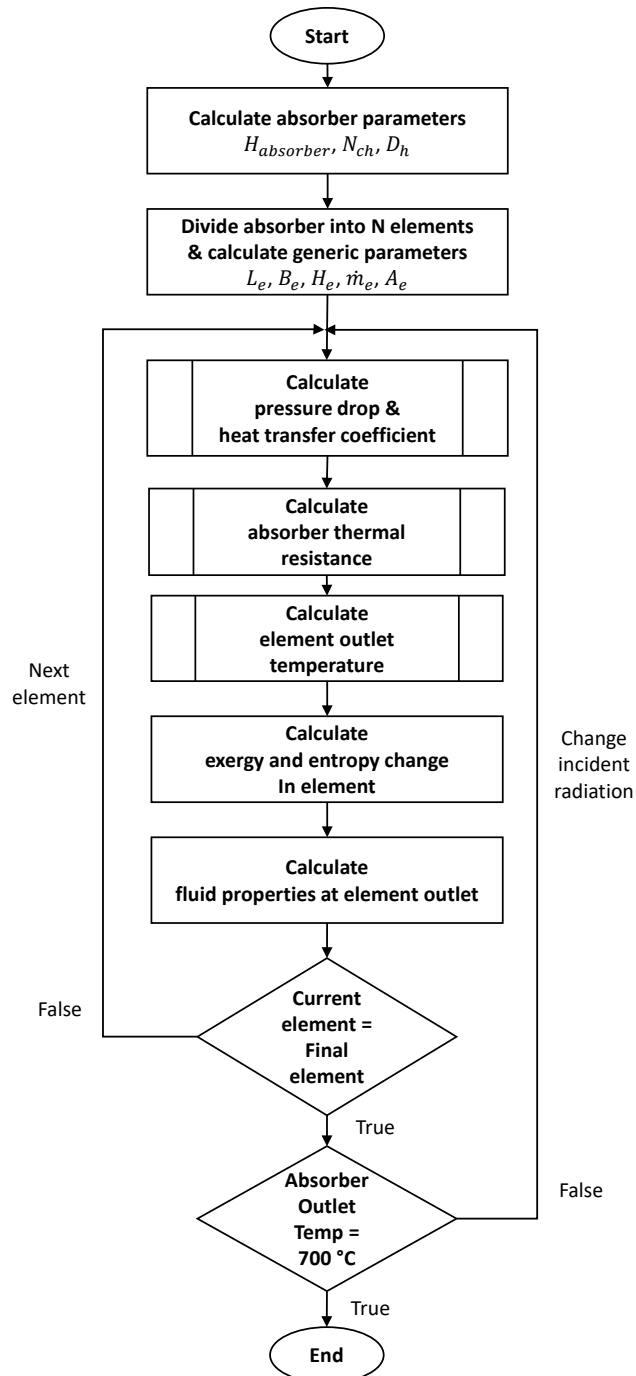


Figure 6. Absorber evaluation subprocess.

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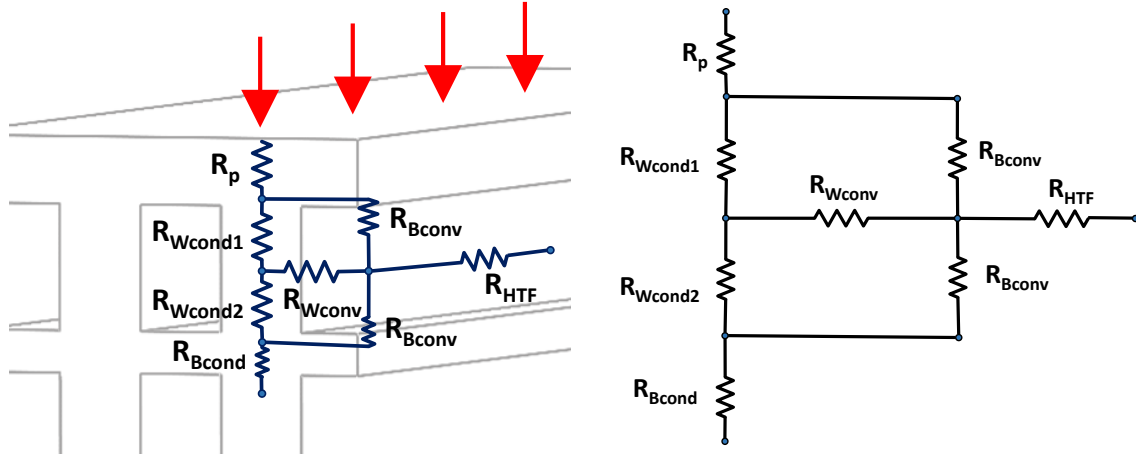
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440

As indicated previously, to quantify the external surface heat losses it is first required to determine the absorber's surface temperature. This temperature distribution through the absorber depth (from the irradiated front surface to the back of the absorber) is determined using a Thermal Resistance Model (TRM) accounting for the fluid flow characteristics and the thermal properties of the receiver material [33],[53],[54]. Figure 7 depicts the thermal model for the simplest geometry, plain rectangular fin, but is also indicative of the other geometries.



441 **Figure 7.** Thermal resistance model in HCE of absorber with plain rectangular fin geometry.
 442 Red arrows indicate the irradiated plane and direction of heat transfer.

443

444 The equation set to determine the thermal resistance for each HCE is further detailed in
 445 equations (7-13). The conductive thermal resistance of the top plate wall (R_p) is defined in
 446 equation (7);

$$R_p = \frac{t_p}{k_{absorber} \left(\frac{B_c}{2}\right) L_e}, \quad (7)$$

447 where t_p is the top plate thickness, $k_{absorber}$ is the thermal conductivity of the absorber and L_e is
 448 the length of the HCE. The convective thermal resistance of the channel walls ($R_{w,conv}$) is given
 449 in equation (8),

$$R_{w,conv} = \frac{1}{h_{conv} H_c L_e}, \quad (8)$$

450 where h_{conv} is the heat transfer coefficient. The convective and conductive thermal resistance of
 451 the base wall ($R_{b,conv}$ and $R_{b,cond}$) are respectively calculated by equations (9) and (10),

$$R_{b,conv} = \frac{1}{h_{conv} \left(\frac{B_c}{2}\right) L_e}, \quad (9)$$

$$R_{b,cond} = \frac{t}{k_{absorber} \left(\frac{B_c + t}{2}\right) L_e}. \quad (10)$$

452 Equation (11) computes the conductive thermal resistance of the channel half wall ($R_{w,cond}$),

$$R_{w,cond} = \frac{H_c}{2 k_{absorber} \left(\frac{t}{2}\right) L_e}. \quad (11)$$

453 The thermal resistance due to the fluid heat gain (R_{HTF}) is defined in equation (12),

$$R_{HTF} = \frac{1}{\rho c_p v A_c}, \quad (12)$$

454 where ρ is the density, c_p is the specific heat and v is the fluid velocity. Referring to Figure 7,
 455 the total thermal resistance of vertically aligned half-channels in a HCE (which can seamlessly
 456 extend to describe the thermal resistance of the whole HCE) is expressed in equation (13):

$$R_{absorber} = R_p + N_{cv} \left[R_{b,cond} + \left\{ R_{b,conv} \parallel \left(R_{w,cond} + \left((R_{w,cond} + R_{b,conv}) \parallel R_{w,conv} \right) \right) \right\} \right]. \quad (13)$$

457 where the parallel symbol (\parallel) between two terms x and y ($x \parallel y$) notates one-half of the
 458 harmonic mean of x and y . As mentioned previously, this thermal resistance model has been
 459 developed specifically for the plain rectangular fin geometry, but it is indicative of all the
 460 compact structures analysed, if the correlations for convection heat transfer coefficient and
 461 friction factor are specified for each of them. These correlations are summarised in the
 462 appendix.

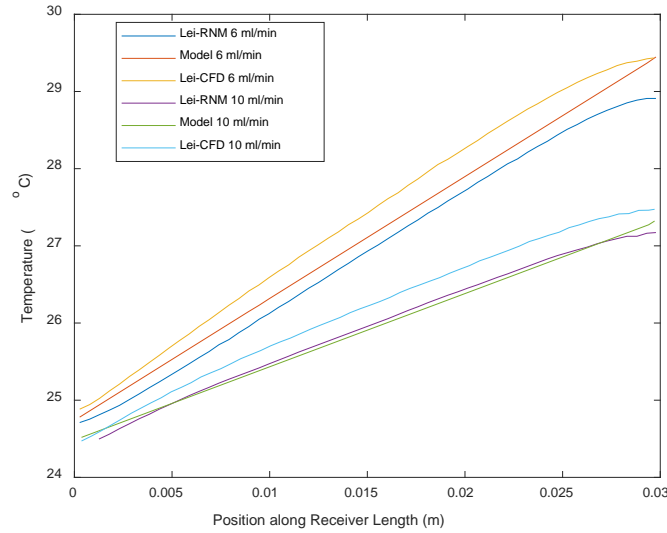
463

464 3.2. Numerical validation of model

465 The thermo-fluid dynamic model has been validated by comparison with data from a
 466 Resistance Network Model (RNM) and a Computational Fluid Dynamics (CFD) model
 467 implemented using the Icepak 4.2 software [54]. The RNM itself was validated using the CFD
 468 model and some limited experimentation. It should be noted that the given application of the
 469 model used for validation was for heat sinks and not specifically for solar receivers. However,
 470 since it uses a single heat flux on one surface and has a multilayer microchannel geometry, it is
 471 well suited for application to solar receiver modelling. The model employed for validation was
 472 used to simulate the behaviour of a plain rectangular fin receiver with three channels; one
 473 horizontal and three verticals. Copper was used as the solid material and a heat flux of 2 W is
 474 applied on the top surface. The model used channels of 0.2 mm \times 0.8 mm with a channel
 475 thickness of 0.2 mm. The overall length, breadth and height were 30 mm, 0.6 mm and 3.2 mm,
 476 respectively. Water was used as fluid with flow rates of 2 ml/min, 6 ml/min and 10 ml/min.
 477 However, results using a flow rate of 2 ml/min were invalid for the validation given the heat
 478 transfer and pressure drop correlations used.

479 As observed in Figure 8, the temperature rise in the current model matches that
 480 predicted by the validation model and the CFD simulation. There is a significant deviation noted
 481 at the beginning and end of the receiver which can be attributed to inherent assumptions made in
 482 both models. The current model assumes a uniform heat flux distribution whereas the validation
 483 model iteratively solves for the heat flux and temperature distribution (finite difference method)
 484 keeping the integrated heat flux over the irradiated surface as constant. The receiver outlet

485 temperature, which may be considered one of the most relevant parameters for receiver's
 486 performance evaluation, is well predicted by the model with deviations from the CFD tool and
 487 validation model less than 2%.
 488



489
 490 **Figure 8.** Temperature evolution comparison between model developed in this work and
 491 resistance network model and CFD simulations given by Lei [54].
 492

493 3.3. Performance indicators and objective functions for the parametric analysis

494 There are several objective functions that can be used when designing and evaluating
 495 solar receivers. These include, but are not limited to, exergy efficiency, thermal efficiency,
 496 optical efficiency and pressure drops [11]. An exergy analysis is presented with the goal of
 497 obtaining a suitable objective function for the receiver optimisation. Such a function must
 498 simultaneously account for the useful fluid heat gain and the undesirable heat losses and
 499 pressure losses. In conventional heat exchanger theory, functions minimising the entropy rise
 500 are widely used [55],[56]. Entropy is generated in the fluid due to the heat gain and also the
 501 pressure drop. The general equation quantifying the entropy addition is given by equation (14):

$$\Delta S_{HTF} = \frac{\dot{q}_{conv,HTF}^2}{\pi k_{avg} T_{avg}^2 Nu_{avg}} + \frac{32 \dot{m}^3 f_D}{\pi^2 \rho_{avg}^2 T_{avg} D_h^5}, \quad (14)$$

502 which is applied to every HCE. In equation (14), the first term on the right hand side is the
 503 contribution made by heat transfer, while the second term is the contribution due to fluid
 504 friction; $\dot{q}_{conv,HTF}$ is the convection heat transfer per unit length of the HCE; \dot{m} is the mass flow
 505 rate per channel, as said before; f_D is the Darcy friction factor; T is the average fluid
 506 temperature; ρ is the average fluid density; k is the average fluid conductivity; Nu is the average
 507 Nusselt number; and D_h is the hydraulic diameter. Alternatively, exergy gain in the fluid may
 508 also be used [57] with the expression for compressible fluids being equation (15)

$$\Delta Ex_{HTF} = \dot{m} \left[\Delta h \left(1 - \frac{T_{amb}}{T_{LMTD}} \right) + R T_{amb} \ln \left(\frac{P_{out}}{P_{in}} \right) \right], \quad (15)$$

509 which is also applied to every HCE; T_{amb} is the ambient temperature; T_{LMTD} is the log mean
 510 temperature difference between the HCE outlet and inlet; R is the ideal gas constant; and P is
 511 the fluid pressure, evaluated at the inlet and outlet of the HCE. Both equation (14) and (15) have
 512 the heat gain term on the left and the pressure drop/fluid friction term on the right. These must
 513 be evaluated at the individual HCE level, and then integrated over the absorber length, as the
 514 independent variables in these equations change continuously. To factor in the receiver heat
 515 losses, the exergy associated to the incident solar radiation is calculated by the Parrot equation
 516 (16) [58]:

$$\Delta Ex_{solar} = \dot{Q}_{solar} \left[1 - \frac{4 T_{amb}}{3 T_{sun}} (1 - \cos \delta)^{1/4} + \frac{T_{amb}}{3 T_{sun}} \right], \quad (16)$$

517 where \dot{Q}_{solar} is the total incident solar radiation on the receiver, also appearing in equation (3);
 518 T_{sun} is the equivalent temperature of the sun as a blackbody (~5,800 K); and δ is the half-angle
 519 of the cone subtended by the sun's disc ($\delta \sim 4.7$ mrad, on a clear day). Combining equations
 520 (15) and (16), a parameter henceforth referred to as the exergy efficiency is obtained as
 521 expressed in equation (17);

$$\eta_{exergy} = \frac{\Delta Ex_{HTF}}{\Delta Ex_{solar}}. \quad (17)$$

522 The exergy efficiency, defined in equation (17), factors in all three effects pertinent to the
 523 performance of a solar receiver. It can hence act as an objective function to each receiver type,
 524 evaluating it for each permutation of the operating parameters, within their ranges, to determine
 525 which is the optimum set for each configuration and overall. As mentioned before, there are also
 526 other performance indicators that are evaluated and presented including energy efficiency [59],
 527 defined in equation (18); and optical efficiency [25], in equation (19);

$$\eta_{energy} = \frac{\dot{Q}_{conv,HTF}}{\dot{Q}_{solar}}, \quad (18)$$

$$\eta_{opt} = \frac{\dot{Q}_{abs}}{\dot{Q}_{solar}}. \quad (19)$$

528

529

530 **4. Results and discussion**

531 *4.1. Parametric study*

532 This section presents the results of the parametric study over the range of the four
533 parameters shown in Table 3: channel height (H_c), channel breadth (B_c), number of vertical
534 channels (N_{cv}), and channel thickness (t). It is important to note that only the most relevant and
535 representative figures are selected, given the large scope and quantum of graphical information.

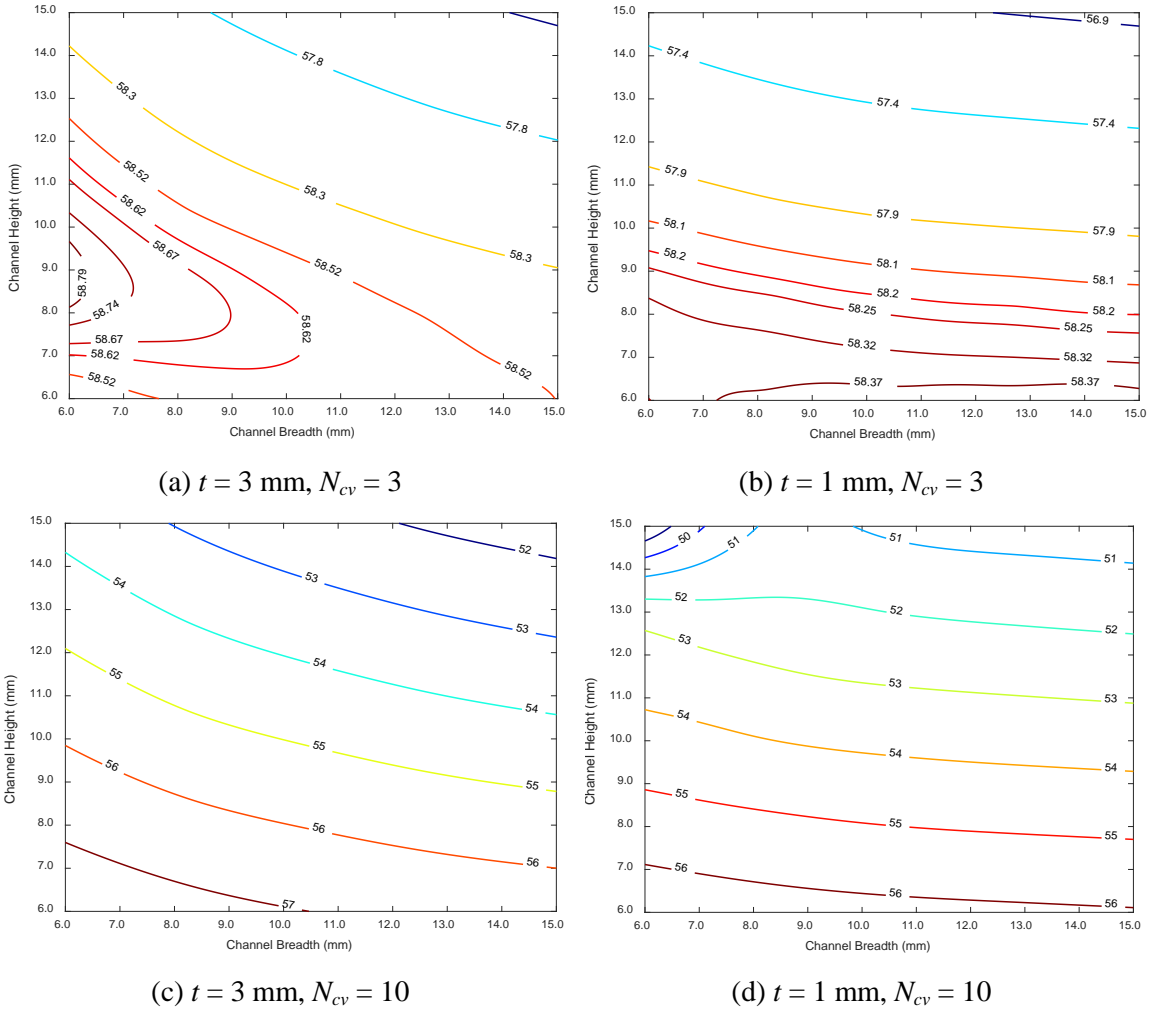
536 The general expected behaviour and performance of the receiver is first elaborated.
537 Increasing the hydraulic diameter, by increasing the channel height and/or the channel breadth,
538 will reduce the fluid flow velocity in the channels. This will adversely affect the convection
539 coefficient and worsen the heat transfer to the fluid. Consequently, more input heat will be
540 required to achieve the same outlet temperature. The thermal resistance of the absorber will also
541 rise causing higher absorber outer surface temperatures and hence increased thermal losses. On
542 the other hand, the decreased fluid velocity reduces the overall pressure drop in the receiver.

543 Besides changes to the hydraulic diameter, the fluid velocity is also affected by the
544 number of channels. For a fixed mass flow rate in the receiver, the increase in the number of
545 channels results in decreasing velocity, as the same flow is divided into more channels. The
546 number of horizontal channels is calculated accounting for the fixed absorber breadth and the
547 variable channel breadth and channel wall thickness. An increase in the number of vertical
548 channels hence causes the increase in the thermal resistance by reducing the convective heat
549 transfer coefficient besides by increasing the number of thermal resistances in series in the
550 absorber network. The pressure drop is also expected to decrease with greater channels as it is
551 indirectly related to channel velocity.

552 Regarding the absorber's equivalent thermal resistance, the channel wall thickness is an
553 important parameter in determining the conductive thermal resistance with thicker walls
554 reducing this resistance and hence allowing for better heat transfer through the solid volume.
555 Thicker walls also reduce the number of horizontal channels and hence, as explained previously,
556 the increase in fluid velocity associated with fewer channels results in higher convection
557 coefficient and lower convective thermal resistance. However, a negative consequence of this
558 effect is that the resulting pressure drop is larger owing to the higher velocities.

559 As observed in Figure 9, the highest exergy efficiency (58.80%) occurs at the largest
560 channel thickness ($t = 3$ mm) and smallest number of vertical channels ($N_{cv} = 3$) within the
561 analysed range. The bettering of the heat transfer, with increased channel thickness and reduced
562 vertical channels, clearly outweighs the increased pressure drop, as marked by the rising exergy
563 efficiency. As it will be seen in Figure 10, the rising channel thickness also favoured the energy
564 efficiency.

565



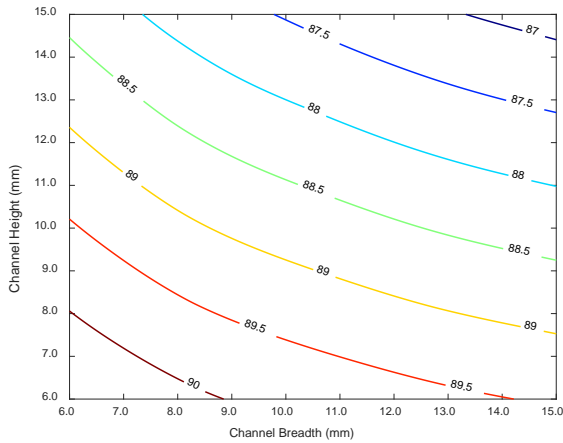
566 **Figure 9.** Exergy efficiency (in %) as function of the channel breadth and height for different
 567 channel thickness and number of vertical channels for the plain rectangular fin geometry. (a)
 568 $t = 3 \text{ mm}, N_{cv} = 3$; (b) $t = 1 \text{ mm}, N_{cv} = 3$; (c) $t = 3 \text{ mm}, N_{cv} = 10$; and (d) $t = 1 \text{ mm}, N_{cv} = 10$.

569

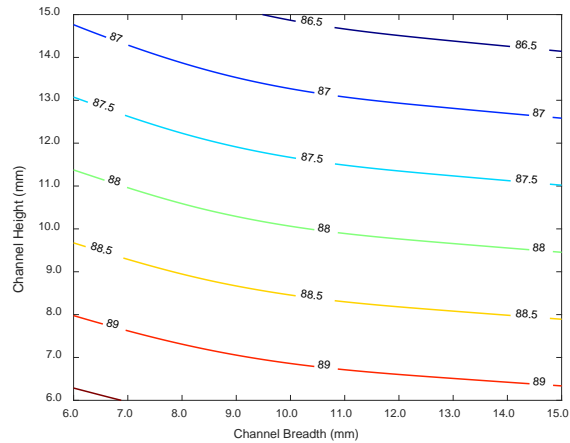
570 In both Figure 9(a) and Figure 9(b), it is observed that rectangular channels are optimal
 571 though with different aspect ratios in each case. This difference in aspect ratios can be attributed
 572 to the competing effects of conduction through the solid channel wall and convection to be the
 573 preferred mode of heat transfer offering the least thermal resistance. Conduction is favourable
 574 when the channel walls are thicker, which leads to the preference for channels with a lower
 575 breadth and hence more horizontal channels and channel walls as seen in Figure 9(a). To keep
 576 the pressure drop in check, the hydraulic diameter must be sufficiently large which entails a
 577 larger channel height to compensate for the optimal channel breadth being at its minimum.

578 The inverse of this phenomenon is seen in Figure 9(b) when the channel thickness is
 579 low and convection offers the less thermally resistive path compared to conduction. Wider
 580 channels reduce the number of horizontal channels and channel walls; shorter channels further
 581 reduce the solid volume and the related conductive thermal resistance.

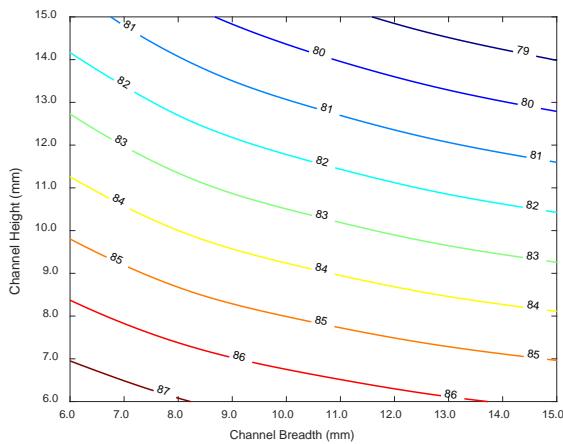
582 When the number of vertical channels is high, as is the case in Figure 9(c) and
 583 Figure 9(d), the channel velocities are so low that the pressure drop factor in the efficiency is
 584 inconsequential. This is further evidenced by the fact that the corresponding energy efficiency
 585 contours i.e Figure 10(c) and Figure 10(d), are identical in trend. The maximum exergy
 586 efficiency hence occurs at the smallest channel sizes with the largest channel thickness which
 587 together provide the least thermal resistance and best heat transfer to the fluid.
 588



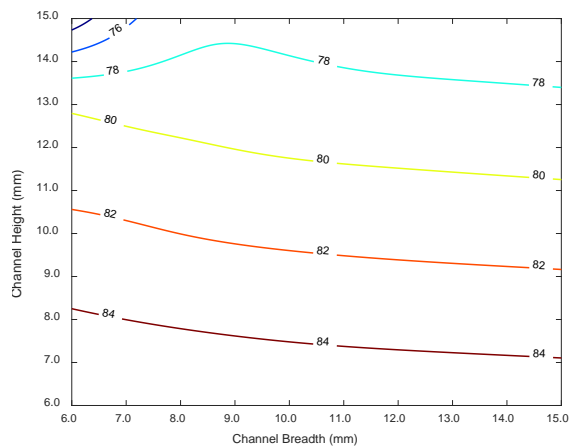
(a) $t = 3 \text{ mm}, N_{cv} = 3$



(b) $t = 1 \text{ mm}, N_{cv} = 3$



(c) $t = 3 \text{ mm}, N_{cv} = 10$



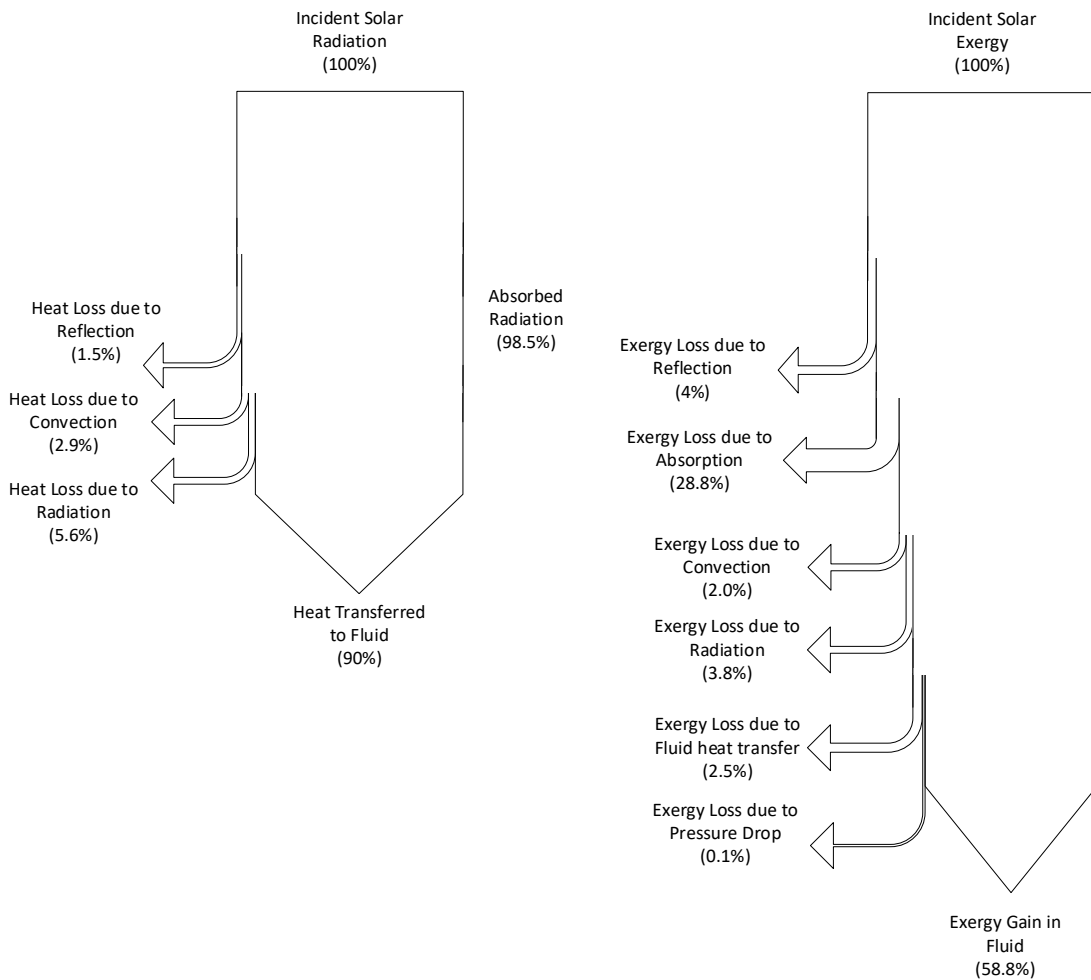
(d) $t = 1 \text{ mm}, N_{cv} = 10$

589 **Figure 10.** Energy efficiency (in %) as function of the channel breadth and height for different
 590 channel thickness and number of vertical channels in the plain rectangular fin geometry. (a)
 591 $t = 3 \text{ mm}, N_{cv} = 3$; (b) $t = 1 \text{ mm}, N_{cv} = 3$; (c) $t = 3 \text{ mm}, N_{cv} = 10$; and (d) $t = 1 \text{ mm}, N_{cv} = 10$.

592

593 In parallel, Figure 10 shows the energy efficiency variation for the same parametric
 594 study applied to the plain rectangular fin geometry. From this, it is evident that the behaviour of
 595 the energy efficiency is relatively simple as it only considers the heat transfer phenomena and
 596 not the related pressure drop. Smaller and fewer channels with thicker walls all work to increase

597 the energy efficiency. These trends hold true for all receiver configurations. On the other hand,
 598 the exergy efficiency behaviour in other receiver configurations is more complex as the HTF
 599 flow characteristics are significantly different.
 600 Sankey and Grassmann diagrams, shown in Figure 11, help visualise the energetic and exergetic
 601 phenomena occurring in the receiver respectively. These diagrams have been generated for the
 602 optimum geometrical configuration of the plain rectangular fin receiver, i.e. the parameter set
 603 resulting in the maximum exergy efficiency.



(a) Sankey (energy flow) diagram

(b) Grassmann (exergy flow) diagram

604 **Figure 11.** Sankey and Grassmann diagram depicting the energy and exergy flow in the plain
 605 rectangular fin geometry receiver with the optimum (maximum exergy efficiency) configuration
 606 set: $H_c = 8.25$ mm, $t = 3$ mm, $N_{cv} = 3$, $B_c = 6$ mm

607

608 Referring to Figure 11(a), the three energy loss mechanisms, namely reflection,
 609 convection and radiation, subtract from incident solar radiation on the receiver as has been
 610 described in section 2.4. The final heat transferred to the fluid, after deducting the energy losses,
 611 represents the same energy efficiency as described by equation (18). There are more physical
 612 phenomena that cause exergy loss in the system, quantified as per standard exergy analyses

613 [60],[61], as can be seen in Figure 11(b). It is observed that the highest exergy loss is associated
614 with the absorption of the incident solar exergy by the receiver. While the exergy loss related to
615 pressure drop, as a fraction of the solar exergy, is negligible it is nonetheless critical to the
616 performance of the receiver as is clear from Figure (9) and the ensuing discussion. The net
617 exergy gain in the fluid corresponds to the exergy efficiency as defined in equation (17).
618

619 4.2. Comparative and optimisation analysis

620 Using the exergy efficiency as the objective function to be maximised, the different
 621 receiver geometries have been optimised (within the operating range of the four varied
 622 parameters) for the configuration yielding the highest exergy efficiency. The energy efficiency
 623 and the pressure drop corresponding to these configurations have also been tabulated and it can
 624 be found in Table 4.

625

626 **Table 4.** Receiver configurations yielding maximum exergy efficiency.

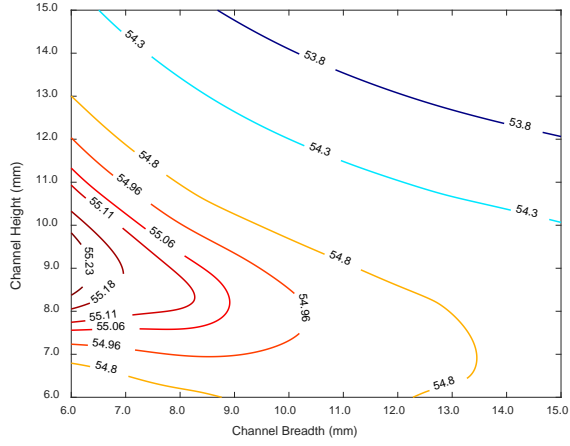
Receiver compact geometry	(Maximum) exergy efficiency (%)	Energy efficiency (%)	Pressure drop (bar)	Absorber panel dimensions			
				H_c (mm)	N_{cv} (-)	B_c (mm)	t (mm)
PRF	58.80	89.96	0.35	8.25	3	6.00	3.00
PTF	55.33	85.43	0.55	6.00	5	6.00	3.00
WF	58.37	89.01	0.24	6.00	5	6.00	3.00
OSF	56.09	86.62	0.57	12.75	3	6.00	3.00
PF	59.21	90.14	0.19	6.00	3	6.00	3.00
LF	53.03	81.82	0.53	7.00	3	7.00	1.00

627

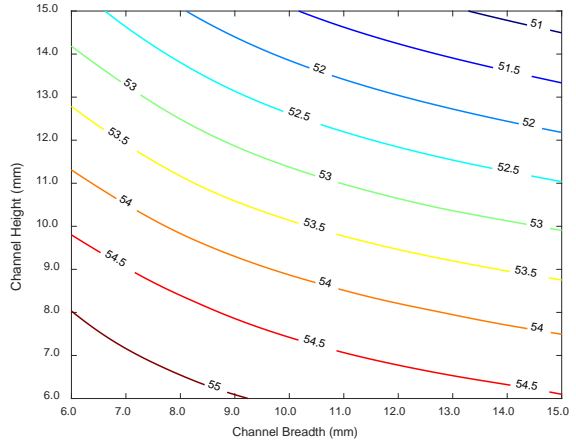
628 As it can be observed from Table 4, the perforated rectangular fin has the highest exergy
 629 efficiency (59.21%), followed by the plain rectangular fin (58.8%) and the wavy fin (58.37%).
 630 The corresponding energy efficiency also follows a similar trend (90.14%, 89.96% and 89.01%
 631 respectively). Owing to the inherent differences in each geometry's heat transfer and pressure
 632 drop characteristics, the resulting optimal configuration for each geometry is different. Simply
 633 put, the exergy efficiency is highest when the heat transfer to the fluid is maximal i.e the heat
 634 losses are minimal and the pressure drop over the flow length is minimal. These two factors run
 635 opposed to one another i.e better fluid heat transfer necessitates greater pressure drop.

636 In this regard there are interesting trade-offs seen between the number of vertical
 637 channels and the channel dimensions, especially its height. This can be clearly visualised in the
 638 contour plots of the different geometries at their optimal configurations in Figure 12 which
 639 excludes the plain rectangular fin geometry to avoid repeating Figure 9(a).

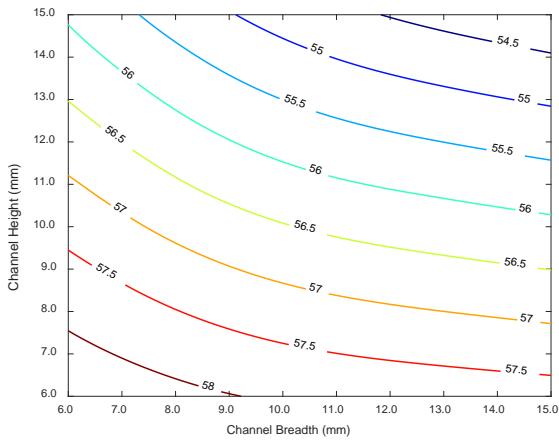
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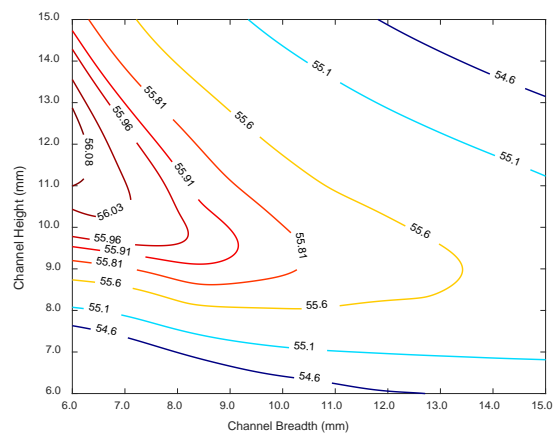
(a) PTF: $t = 3$ mm, $N_{cv} = 3$



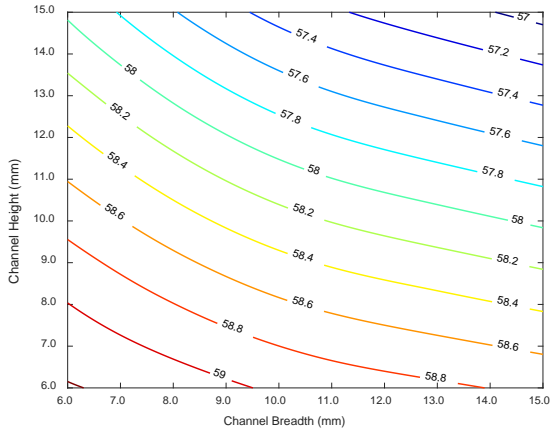
(b) PTF: $t = 3$ mm, $N_{cv} = 5$



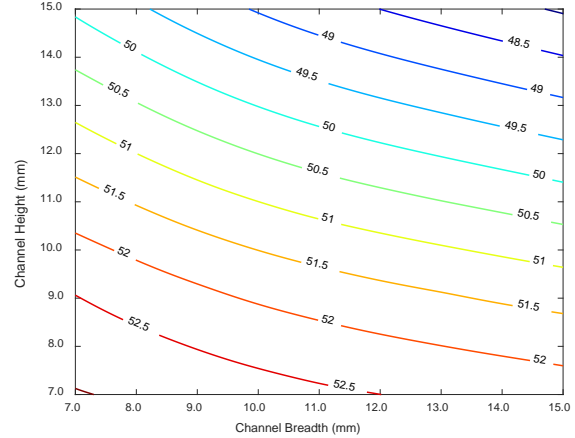
(c) WF: $t = 3$ mm, $N_{cv} = 5$



(d) OSF: $t = 3$ mm, $N_{cv} = 3$



(e) PF: $t = 3$ mm, $N_{cv} = 3$



(f) LF: $t = 1$ mm, $N_{cv} = 3$

641 **Figure 12.** Exergy efficiency as function of the channel breadth and height, for different
 642 absorber geometries. (a) PTF: $t = 3$ mm, $N_{cv} = 3$; (b) PTF: $t = 3$ mm, $N_{cv} = 5$; (c) WF: $t = 3$ mm,
 643 $N_{cv} = 5$; (d) OSF: $t = 3$ mm, $N_{cv} = 3$; (e) PF: $t = 3$ mm, $N_{cv} = 3$; (f) LF: $t = 1$ mm, $N_{cv} = 3$.

644
 645

646 Comparing the plain triangular fin geometry's optimal configuration in Figure 12(b)
647 with its performance using fewer vertical channels in Figure 12(a), it can be seen that the trends
648 in both ultimately serve to reduce the pressure drop though in different ways. By either
649 increasing the number of channels, as is the case with the wavy fin in addition to the plain
650 triangular fin, or the hydraulic diameter, which is the case with the plain rectangular and offset
651 strip fin, this purpose may be sufficiently served.

652 In the case of the perforated fin, the pressure drop is sufficiently low to allow for the
653 configuration with, theoretically, the greatest pressure drop to coincide with the optimal exergy
654 efficiency operation point. The louvred fin geometry has the interesting feature of combining
655 higher convective heat transfer with greater pressure loss and this is what leads to its optimal
656 configuration having the smallest channel thickness. However, this excessive pressure loss
657 causes it to have the worst exergy efficiency.

658 It should be noted though that the relative differences between the various CHE optimal
659 configurations (especially the aforementioned top three performers) are not large and may fall
660 within the range of modelling uncertainty. In this regard, the correlations in the appendix
661 already have uncertainties in the range of 3-10%. While the precise values of these performance
662 indicators may be further refined and their errors ascertained, these results provide a good
663 indication of the relative performances of the different receiver internal flow geometries. They
664 also highlight the immense scope of work in this area and the importance of a thorough and
665 careful optimisation analysis paying heed to the selection of objective functions and figures of
666 merit.

667

668 **5. Conclusions and future work**

669 Compact heat exchangers are a commercially demonstrated technology that improves
670 the heat transfer and the volumetric efficiency of heat exchange devices. Such compact heat
671 exchangers come in many geometrical forms, when it comes to the internal channels or flow
672 paths, each with their unique properties. The application of these concepts to solar towers
673 results in microchannel receivers that show the potential of operating at high temperatures while
674 maintaining high reliability and thermal efficiency. This is especially true when the heat transfer
675 fluid employed is a pressurised gas, as smaller channels are thermo-mechanically more capable
676 of handling such fluids. In this context, this study has investigated the use of different receiver
677 internal flow geometries, inspired by compact heat exchanger concepts, to analyse the
678 performance of various microchannel receivers. It has been assumed that the heat transfer fluid
679 through the receiver is pressurised air. In this analysis, the particular conditions at the receiver
680 inlet/outlet are determined by coupling the receiver to a supercritical CO₂ recompression cycle,
681 although other coupling possibilities would also be valid.

682 For this microchannel receiver, the compact geometries analysed were the plain
683 rectangular fin, plain triangular fin, perforated fin, wavy fin, offset strip fin and the louvred fin.
684 Besides comparing several compact geometries, internal parametric and optimisation studies
685 were performed with each flow geometry to determine the optimum configuration. The
686 parameters varied were the channel height, breadth, wall thickness (between channels) and the
687 number of vertical channels (number of channels along the height dimension).

688 Exergy efficiency has been defined and identified as a suitable performance indicator
689 and objective function to be maximised for the optimisation study. It is deemed suitable as it
690 accounts for the heat losses besides the heat transferred to the fluid and the pressure drop across
691 the receiver. Perforated fin followed by plain rectangular and wavy fin receivers were identified
692 as the best performing receiver subtypes.

693 The thermal resistance of the receiver, in addition to the pressure drop, plays an
694 important role in determining the optimal geometric configuration. For the best heat transfer to
695 the fluid, which is an important part of the exergy efficiency, the smallest channels or lowest
696 hydraulic diameters are preferred. If this causes excessive pressure drops, either deeper channels
697 or a greater number of vertical channels is preferred to improve the exergy efficiency by
698 mitigating the pressure drop. In virtually all cases narrower channels with thicker walls are
699 favoured because of the better conduction through the solid receiver channel walls compared to
700 the parallel heat flow path of convection via the pressurised air. The lower thermal resistance
701 lowers the receiver's heat loss, as well as provides a more uniform temperature through the
702 receiver.

703 The methodology used in the analysis, its inherent assumptions in addition to the
704 operating and boundary conditions and limits, lends itself to the characteristics of gas phase

705 receivers and the unique challenges posed in studying such receivers. The selection of operating
706 and boundary conditions including, but not limited to, parameters such as the channel velocity,
707 view factor and incident flux play an important part in the receiver's performance and optimal
708 configuration. Investigating the physical limits and phenomena limiting the operation boundary
709 of gas receivers, aside from developing methodologies for their analysis, appears as an
710 interesting area of study.

711 The present results indicate a promising scope to the use of compact heat exchanger
712 concepts for solar receivers especially with regards the internal flow channel geometry. While
713 the results themselves carry some uncertainties (an area of future investigation), this analysis
714 clearly demonstrates the utility of using exergy efficiency as a performance indicator and it
715 provides indications to the comparative performance of different receiver geometry types. A
716 regression analysis and modelling for each geometry is proposed to allow for further analysis
717 and its easier coupling in other studies including overall cycle analyses. More work is required
718 to more accurately model the thermo-physical processes occurring in such microchannel
719 receivers. 3D Computational Fluid Dynamics (CFD) models and experimental demonstrations
720 of these microchannel receivers are also required to validate these findings.

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726

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951

952 **Appendix**

953 *A. 1. Compact heat exchanger type structures applied to central solar receivers*

954 There is no single definition of a Compact Heat Exchanger (CHE) although it is usually
 955 thought of as a Heat Exchanger (HX) having a surface area density above $700 \text{ m}^2/\text{m}^3$ or a
 956 hydraulic diameter below 6 mm, if at least one fluid is a gas [27],[62],[63]. The reduction in
 957 hydraulic diameter leads to the following outcomes:

- 958 • A reduction in the solid volume required for the same effective heat transfer area potentially
 959 resulting in significant savings in material costs [64].
- 960 • A higher heat transfer coefficient, due to decreased diffusion length compared to
 961 conventional channels or tubes, as mentioned before [65].
- 962 • The main drawback of this concept which is that pressure drop increases, although this
 963 effect may be offset if the gas is pressurised, as velocity is much lower at high pressure, for
 964 the same mass flow and same cross section, as explained in the previous section.

965 There have been several studies investigating the use of CHEs as the intermediary HX
 966 between the solar field and the power block [42],[66],[67]. but fewer studies into directly using
 967 CHE geometries and concepts for solar receivers, as summarised in the next paragraphs.

968 The operational limits of the main types of CHEs are presented in **Table A.1**, although
 969 it should be noted that these limits are not absolute and largely depend on the materials used and
 970 manufacturing processes. It is intended to be indicative of the relative capacities of each type of
 971 CHE.

972

973 **Table A.1.** Operational limits of CHEs. PHE represents Plate Heat Exchanger; PFHE stands for
 974 Plate Fin Heat Exchanger; PCHE denotes Printed Circuit Heat Exchanger; and SHE denotes
 975 Spiral Heat Exchanger. [27]

976

CHE Type	Maximum temperature (°C)	Maximum pressure (bar)
Gasketed PHE	200-250	35
Brazed PHE	225	45
PFHE	800	120
Diffusion bonded PFHE	800	620
PCHE	900	500-1,000
SHE	540	25

977

978 Given the high temperature requirements of all solar receivers ($> 900\text{ }^{\circ}\text{C}$), Plate Heat
979 Exchangers (PHEs) are not feasible and the CHEs must use materials that can withstand such
980 temperatures such as ceramics, nickel and titanium alloys. Diffusion bonded Plate Fin Heat
981 Exchangers (PFHEs) and Printed Circuit Heat Exchangers (PCHEs) were considered the most
982 suitable candidates for application as solar receivers, due to their high efficiency as well as
983 mechanical strength. Nevertheless, for typical PGR working pressures, even in the case of direct
984 coupling with a supercritical power cycle (approximately 200 bar), it is sufficient to use a
985 diffusion bonded PFHE type, so the research has focused on this type, yielding to several
986 prototypes described in next section.

987 Finally, it is interesting to note that there are additive methods to manufacture these
988 compact structures including Electrically Assisted (EA) forming [68] and Selective Laser
989 Melting (SLM) [28] that provide a greater degree of flexibility in the design of the CHE
990 microchannels. There are also novel techniques, specifically Electrical Discharge Machining
991 (EDM), employed to increase the aspect ratio (channel height to width) in PFHEs [35],[36].
992

993 A. 2. Heat transfer correlations (channel flow)

994 Heat transfer correlations, in terms of Colburn factor (j) or Nusselt Number (Nu), are presented in Table A.2, for the different compact geometries
 995 analysed in this work, at different operating conditions.

996

997

Table A.2. Heat transfer correlations, in terms of Colburn factor (j) or Nusselt Number (Nu) for the different compact geometries analysed

Receiver	Validity	Correlation	Reference
Plain Rectangular Fin	$2700 < Re_{D_h} < 10100$	$j = 0.609 Re_{D_h}^{-0.493} \left(\frac{t}{H_c}\right)^{-0.011} \left(\frac{p}{H_c}\right)^{-0.071} \left(\frac{L_e}{D_h}\right)^{-0.298}$	[69],[70]
	$Re_{D_h} > 10100$ $0.5 < Pr_{D_h} < 2000$	$Nu_{D_h} = \frac{f_D}{2} (Re_{D_h} - 1000) Pr_{D_h} \left(1 + 12.7 \left(\frac{f_D}{2}\right)^{0.5} \left(Pr_{D_h}^{\frac{2}{3}} - 1\right)\right)^{-1} \left(1 + \left(\frac{D_h}{L_e}\right)^{\frac{2}{3}}\right)$	[71],[72]
Plain Triangular Fin	$Re_{D_h} > 2300$ $0.5 < Pr_{D_h} < 2000$	$Nu_{D_h} = \frac{f_D}{2} (Re_{D_h} - 1000) Pr_{D_h} \left(1 + 12.7 \left(\frac{f_D}{2}\right)^{0.5} \left(Pr_{D_h}^{\frac{2}{3}} - 1\right)\right)^{-1} \left(1 + \left(\frac{D_h}{L_e}\right)^{\frac{2}{3}}\right)$	[71],[72]
Wavy Fin	$Re_{D_h} < 1900$	$j = 0.2951 Re_{D_h}^{-0.1908} \left(\frac{p}{D_h}\right)^{0.7356} \left(\frac{H_c}{D_h}\right)^{0.1378} \left(\frac{t}{D_h}\right)^{0.0485} \left(\frac{2A}{D_h}\right)^{0.2467} \left(\frac{L}{D_h}\right)^{-0.4976}$	[73]
	$Re_{D_h} > 1900$	$j = 0.7293 Re_{D_h}^{-0.3637} \left(\frac{p}{D_h}\right)^{0.7966} \left(\frac{H_c}{D_h}\right)^{0.2398} \left(\frac{L}{D_h}\right)^{-0.4979} \left(\frac{t}{D_h}\right)^{0.0402} \left(\frac{2A}{D_h}\right)^{0.2012} \left(\frac{L_e}{D_h}\right)^{-0.3026}$	
Offset strip Fin	$Re_{D_h} < 2000$	$j = 1.37 Re_{D_h}^{-0.67} \left(\frac{L_s}{D_h}\right)^{-0.25} \left(\frac{B_c}{H_c}\right)^{-0.184}$	[74]
	$Re_{D_h} \geq 2000$	$j = 1.17 Re_{D_h}^{-0.36} \left(\frac{L_s}{D_h} + 3.75\right)^{-1} \left(\frac{t}{D_h}\right)^{0.089}$	
Louvred (Triangular) Fin		$j = 0.65842 \left(\frac{L_e}{L_l Re_L Pr_{D_h}}\right)^{0.6317} \left(\frac{L_l}{H_c}\right)^{-0.4825} \left(\frac{L_e \tan(L_a)}{H_c}\right)^{-0.433} \left(\frac{L_s}{t}\right)^{-1.1902}$	[75]
Perforated (Rectangular) Fin	Nu_{SF} (solid fin) is calculated as is done with Plain Rectangular Fin		[76],[77]
	$P_d > 0.04$	$Nu_{PF} = Nu_{SF} 1.296 Re_{DSF}^{-0.0357} (1 - P)^{0.269}$	
	$P_d < 0.04$	$Nu_{PF} = Nu_{SF} (0.0307 Re_{DSF}^{0.226} + 0.583(1 - P)^{0.704})$	

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999 A. 3. Pressure drop correlations (channel flow)

1000 Pressure drop correlations (in terms of f_D or f_F) are available for the different geometries at different operating conditions. The most relevant ones for
 1001 this study are presented in Table A.3:

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Table A.3. Pressure drop correlations in terms of Darcy friction factor (f_D) or Fanning friction factor (f_F) for the different compact geometries analysed.

Receiver	Validity	Correlation	Reference
Plain Rectangular Fin	$2700 < Re_{D_h} < 10100$	$f_F = 0.059 Re_{D_h}^{-0.117} \left(\frac{t}{H_c}\right)^{0.118} \left(\frac{p}{H_c}\right)^{-0.253} \left(\frac{L_e}{D_h}\right)^{-0.147}$	[69],[78]
	$Re_{D_h} > 10100$ $0.5 < Pr_{D_h} < 2000$	$f_D = (1.82 \log_{10}(Re_{D_h}) - 1.64)^{-2}$	[71],[72]
Plain Triangular Fin	$Re_{D_h} > 2300$ $0.5 < Pr_{D_h} < 2000$	$f_D = (1.82 \log_{10}(Re_{D_h}) - 1.64)^{-2}$	[71],[72]
Wavy Fin	$Re_{D_h} < 1900$	$f_F = 38.7488 Re_{D_h}^{-0.3840} \left(\frac{p}{D_h}\right)^{-1.479} \left(\frac{H_c}{D_h}\right)^{-0.3696} \left(\frac{L}{D_h}\right)^{-1.4542} \left(\frac{t}{D_h}\right)^{0.1016} \left(\frac{2A}{D_h}\right)^{1.0903} \left(\frac{L_e}{D_h}\right)^{-0.1549}$	[73]
	$Re_{D_h} > 1900$	$f_F = 52.2375 Re_{D_h}^{-0.3524} \left(\frac{p}{D_h}\right)^{-1.6277} \left(\frac{H_c}{D_h}\right)^{-0.3529} \left(\frac{L}{D_h}\right)^{-1.7484} \left(\frac{t}{D_h}\right)^{0.1034} \left(\frac{2A}{D_h}\right)^{1.2294} \left(\frac{L_e}{D_h}\right)^{-0.2371}$	
Offset strip Fin	$Re_{D_h} < 2000$	$f_F = 5.55 Re_{D_h}^{-0.67} \left(\frac{L_s}{D_h}\right)^{-0.32} \left(\frac{B_c}{H_c}\right)^{-0.092}$	[74]
	$Re_{D_h} \geq 2000$	$f_F = 0.83 Re_{D_h}^{-0.20} \left(\frac{L_s}{D_h} + 0.33\right)^{-0.5} \left(\frac{t}{D_h}\right)^{0.534}$	
Louvred (Triangular) Fin		$f_F = 0.07667 \left(\frac{L_e}{L_l Re_L Pr_D}\right)^{0.3211} \left(\frac{L_l}{H_c}\right)^{-2.0217} (\tan(L_a))^{-2.3501} \left(\frac{L_s}{t}\right)^{-2.5343}$	[75]
Perforated (Rectangular) Fin	The solid fin pressure drop is calculated as was the Plain Rectangular Fin's and its ratio with perforated fins is taken as: $\Delta p_{PF} = \Delta p_{SF} (0.97 - Re_{DSF} 10^{-5})$		[76],[77]

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